

RUB

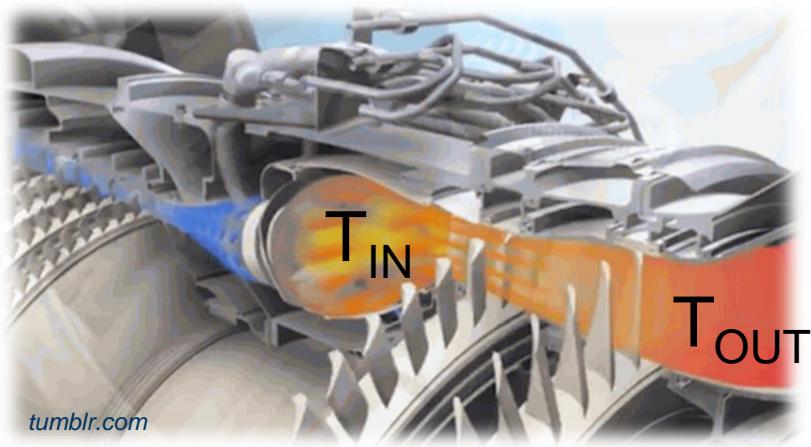
RUHR-UNIVERSITÄT BOCHUM

ADVANCED HEAT-TREATMENTS OF TURBINE BLADE MATERIALS



Lehrstuhl
Werkstofftechnik
Materials Technology

Thermal efficiency of heat engines



tumblr.com

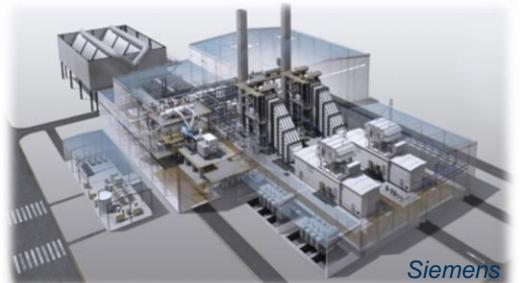
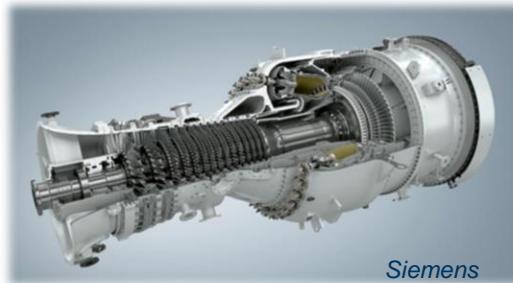
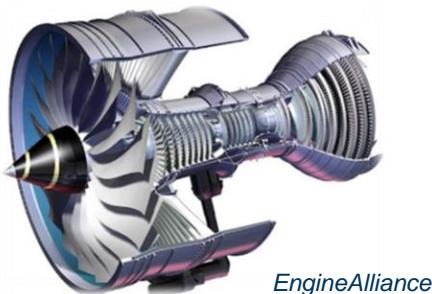
$$\eta_c = 1 - \frac{T_{OUT}}{T_{IN}}$$

- Carnot cycle efficiency: highest theoretical efficiency for the conversion of heat into mechanical energy

Gas turbines

- Aerospace:
 - Airplanes / Helicopters
 - Engine

- Energy sector:
 - Gas / Steam Power plants
 - Power supply



Gas turbines: Geometries of turbine blades

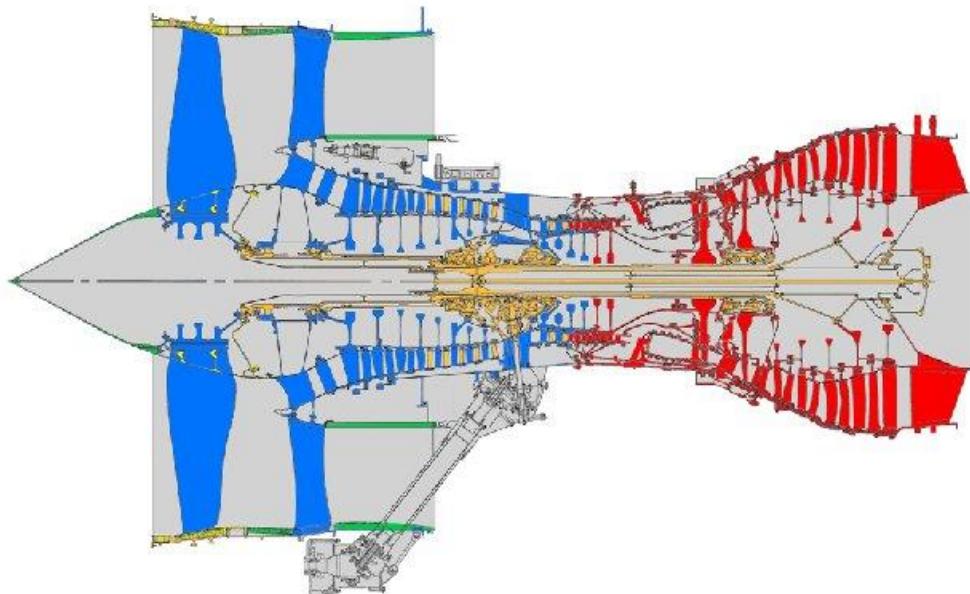


- Geometry is directly dependent of:
 - Location within the Gas Turbine
 - Temperature and load regime



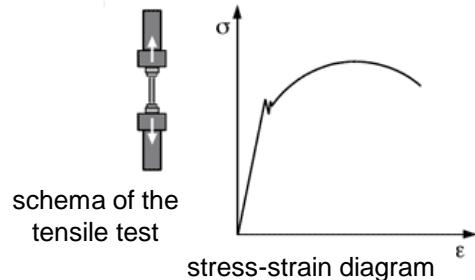
Material groups in gas turbines

- Titanium and its alloys: High strength, low density
- Steels (static parts of the compressor)
 - Martensitic 9-12% Cr steels
 - Austenitic Cr-Ni steels
- Superalloys ($T > 0.7 T_{sol}$)
 - Nickel
 - Cobalt
 - Iron-Nickel
- Aluminum alloys
- Magnesium alloys



<http://www.phase-trans.msm.cam.ac.uk/2003/Superalloys/coatings/materials.html>

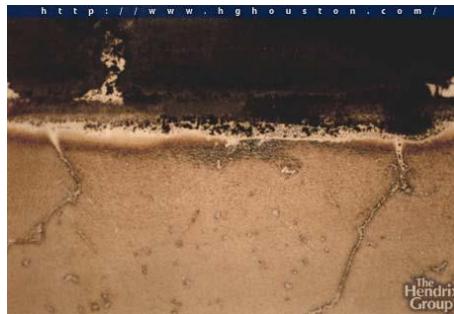
Types of loads on turbine blades:



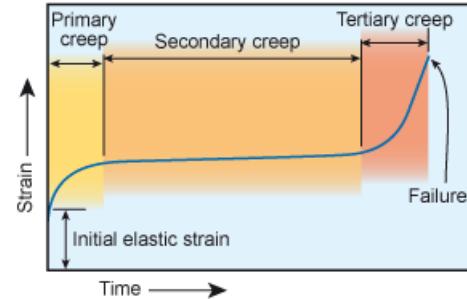
schema of the tensile test

stress-strain diagram

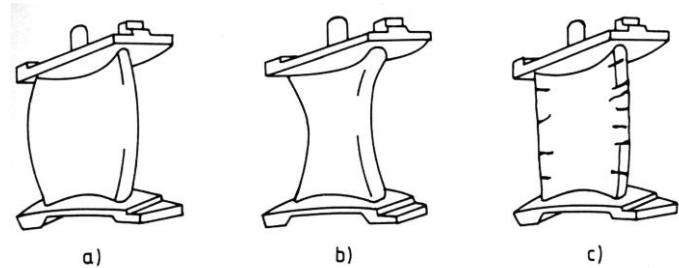
High-temperature strength



HT- corrosion resistance



Creep resistance

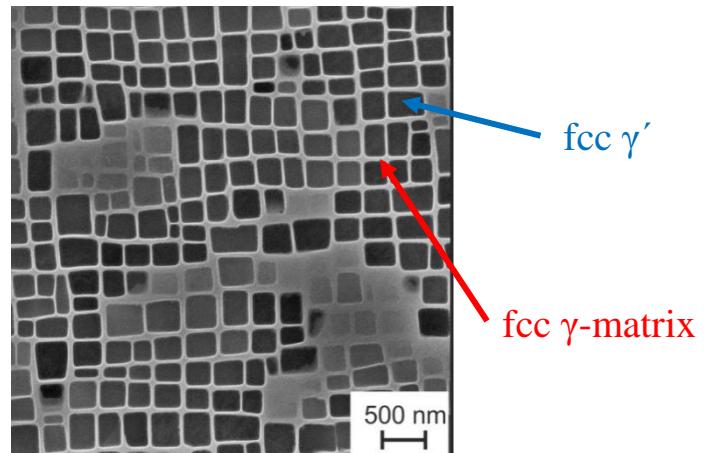


Thermal shock resistance

Blades behind the combustion chamber

SINGLE-CRYSTAL (SX) Ni base superalloys

- Excellent thermal and mechanical properties at $\uparrow T$

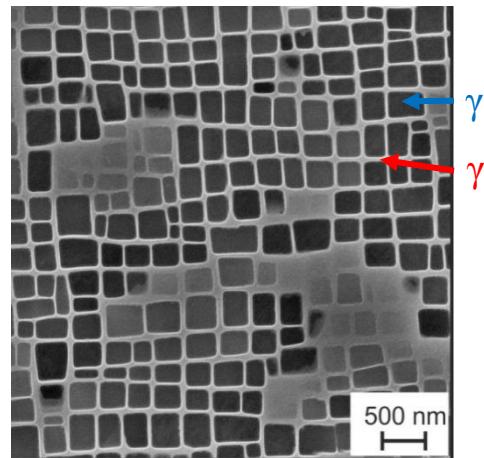


Animation content from: <https://youtu.be/wYHch5QIWTQ>

SINGLE-CRYSTAL (SX) Ni base superalloys



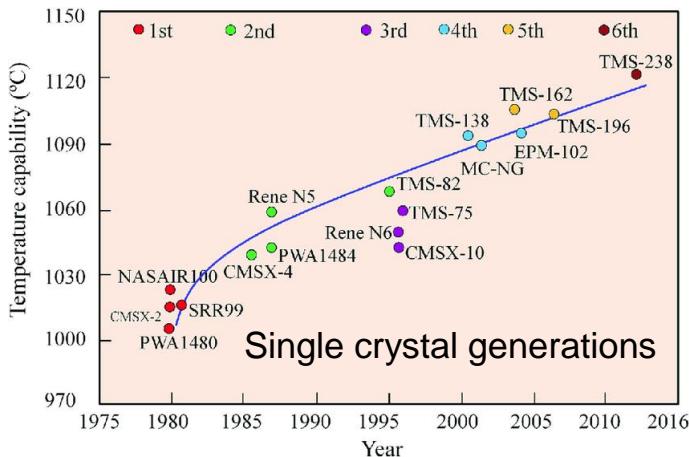
Make Magazine



- Technical single crystal
 - Two-phase microstructure
- No grain boundaries
- Nickel: High melting point
- **Strengthening mechanisms:**
 - Solid solution strengthening
 - Particle hardening through coherent γ' precipitates

wt.%	Co	Al	Cr	W	Ti	Ta	Re	Hf	Mo	Ni
ERBO/1	9.7	5.6	6.4	6.4	1.0	6.5	3.0	0.1	0.6	Bal.

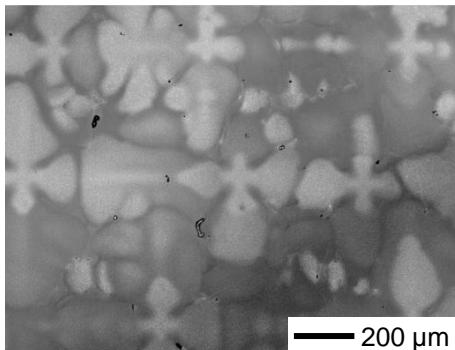
SINGLE-CRYSTAL (SX) Ni base superalloys



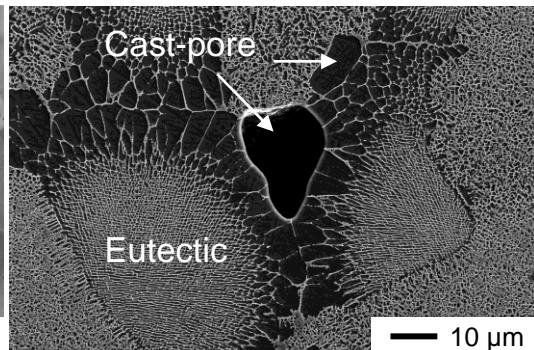
H. Long et al. Journal of Alloys and Compounds 743 (2018)

SX generation	Re content	Refractory content = Mo+Ta+W+Re
1 st	0 wt.%	14 wt.%
2 nd	3 wt.%	16.5 wt.%
3 rd	6 wt.%	20 wt.%
4 th+ Ru !	...

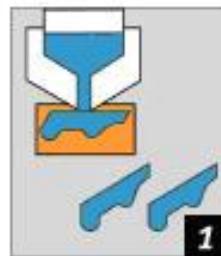
CMSX-10K
Strong segregation



Interdendritic region (ID)



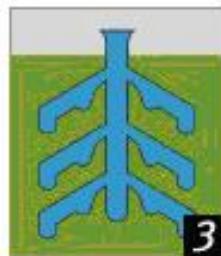
SX manufacturing: precision casting



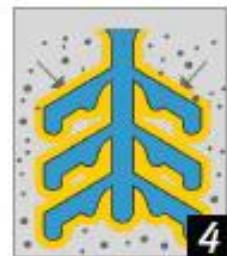
Wax Pattern



Tree-Making



Slurry Dipping



Coating



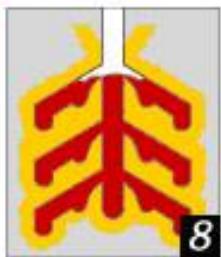
Ceramic Shell Cluster



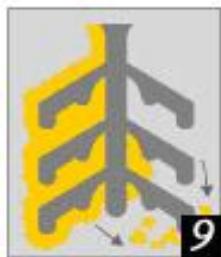
Dewaxing



Burn-out and Preheating



Casting



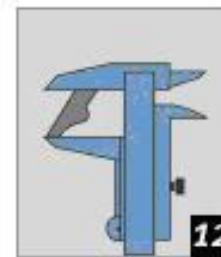
Shell-Removal



Cutting



Gate Cut-off



Inspection

SX manufacturing: precision casting

- From wax model to ceramic shell mold



Tree making



Slurry dipping and coating



SX manufacturing: precision casting

- From Burn-out to casting



Burn-out and pre-heating the shell

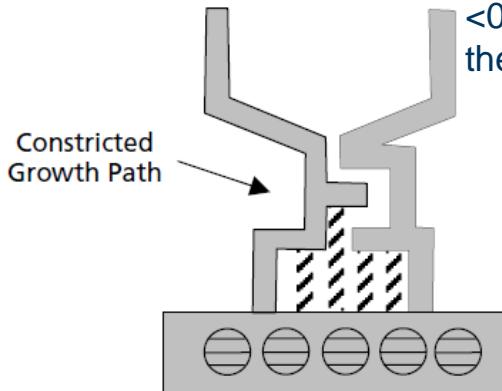
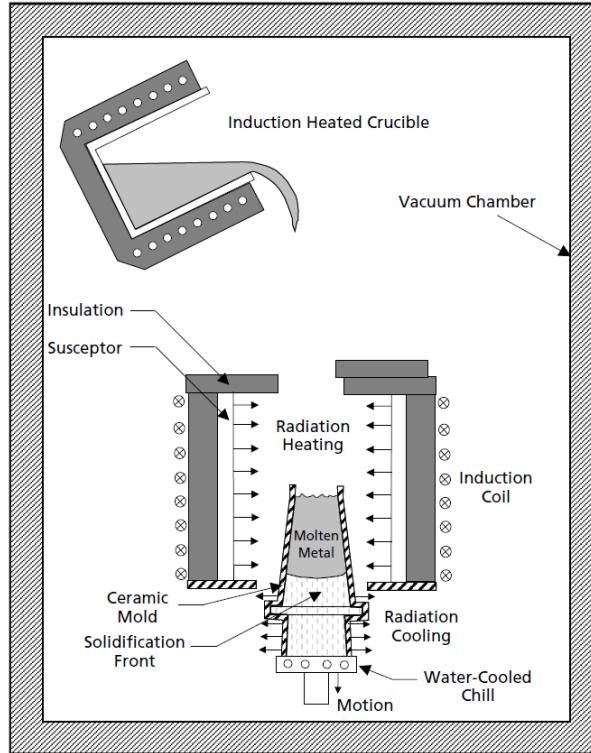


Casting in Vacuum



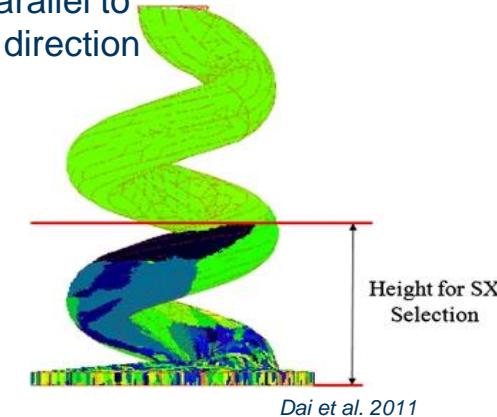
Shell removal

Vacuum investment casting: Bridgman process



High Rate Solidification

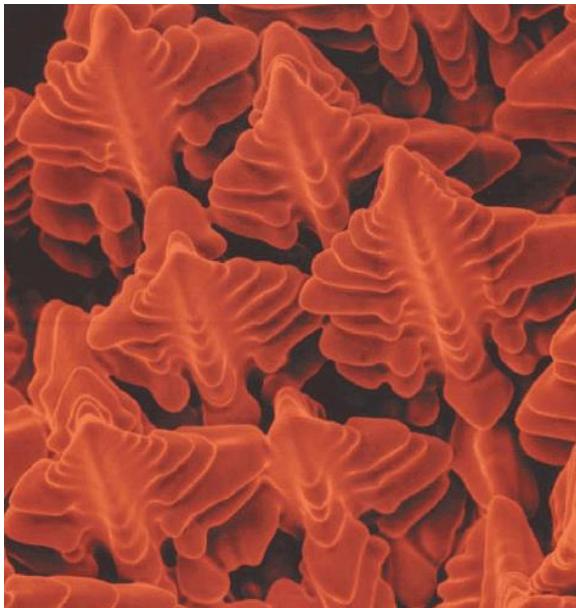
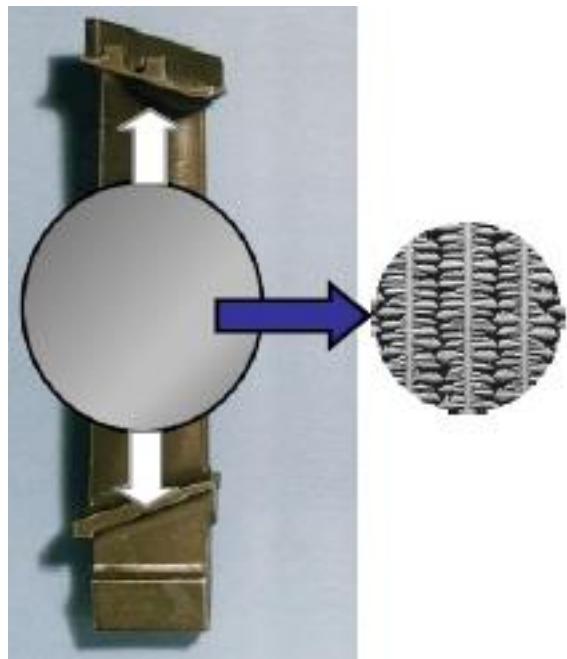
- Strong T gradient:
 - 70-130 K/cm
- Withdrawal rate:
 - 30 cm/h



Grain selector (helix)

- Principle: On the cooling plate polycrystalline solidification, only one grain may grow

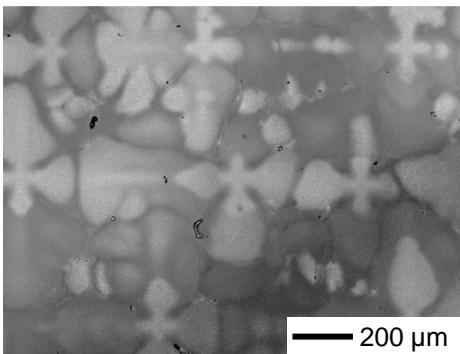
Vacuum investment casting: Dendritic solidification



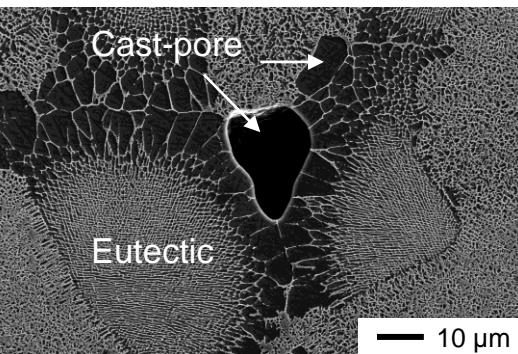
Vacuum investment casting: Dendritic solidification

- Inhomogeneities / defects

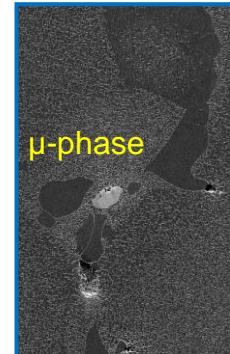
Strong chemical segregation



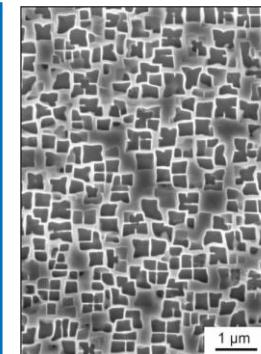
Interdendritic region (ID)



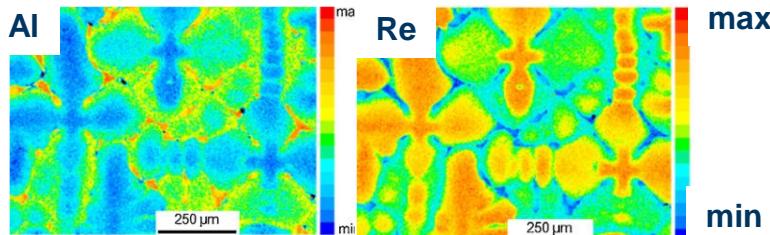
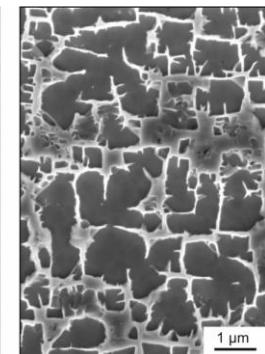
TCP – phases



Dendritic



Interdendritic



Parsa et al. 2015

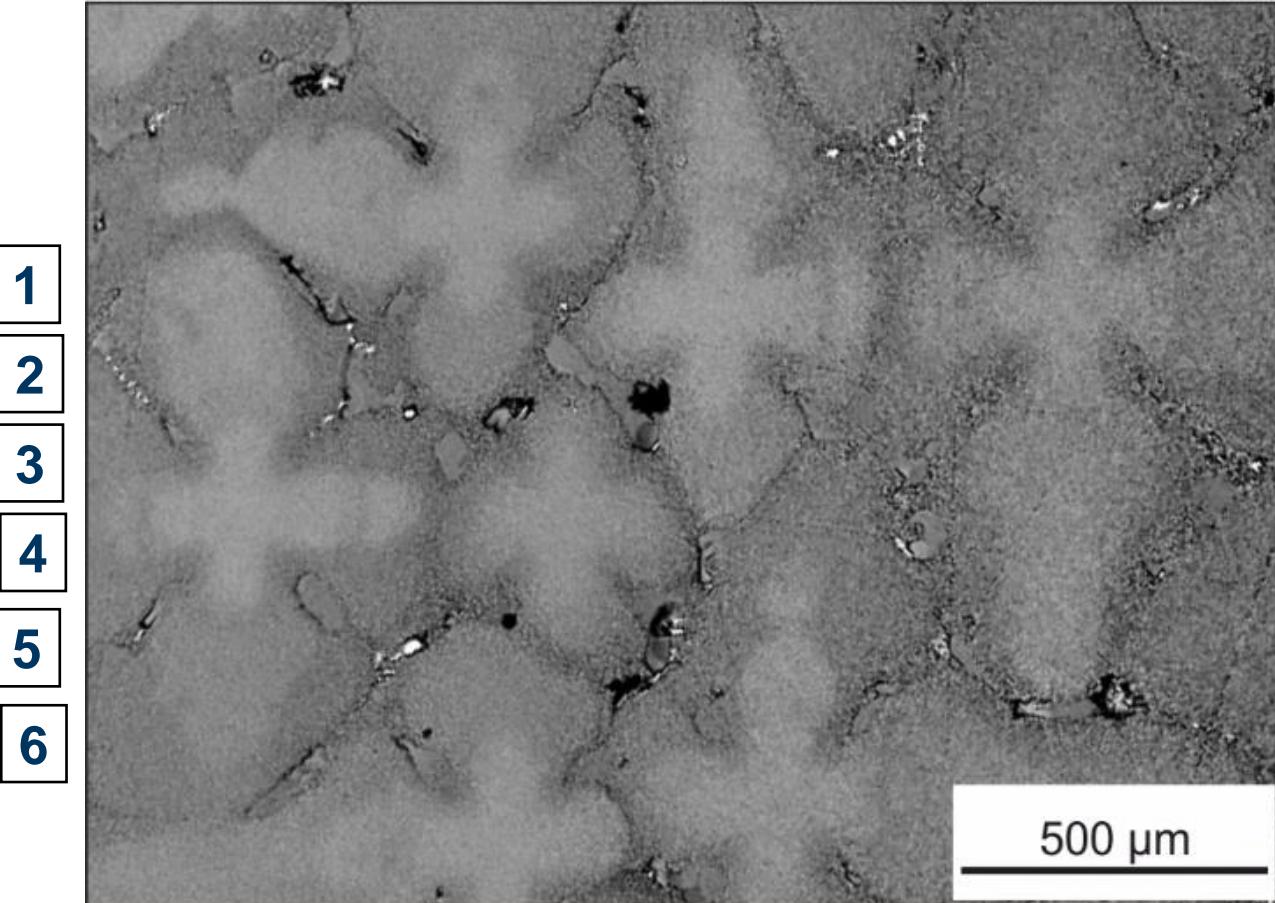
- Large-scale (chemical) heterogeneities: Dendritic structure, chemical segregation, **eutectics**, TCP, pores
- Small-scale heterogeneity: γ/γ' microstructure

Exercise Nr.1

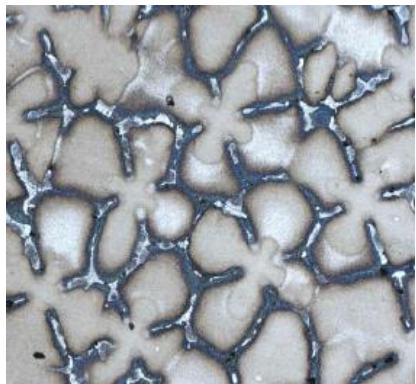
- Position:

1. Dendrite core
2. Dendrite arm
3. Interdendritic
4. Pore
5. Eutectic
6. TCP

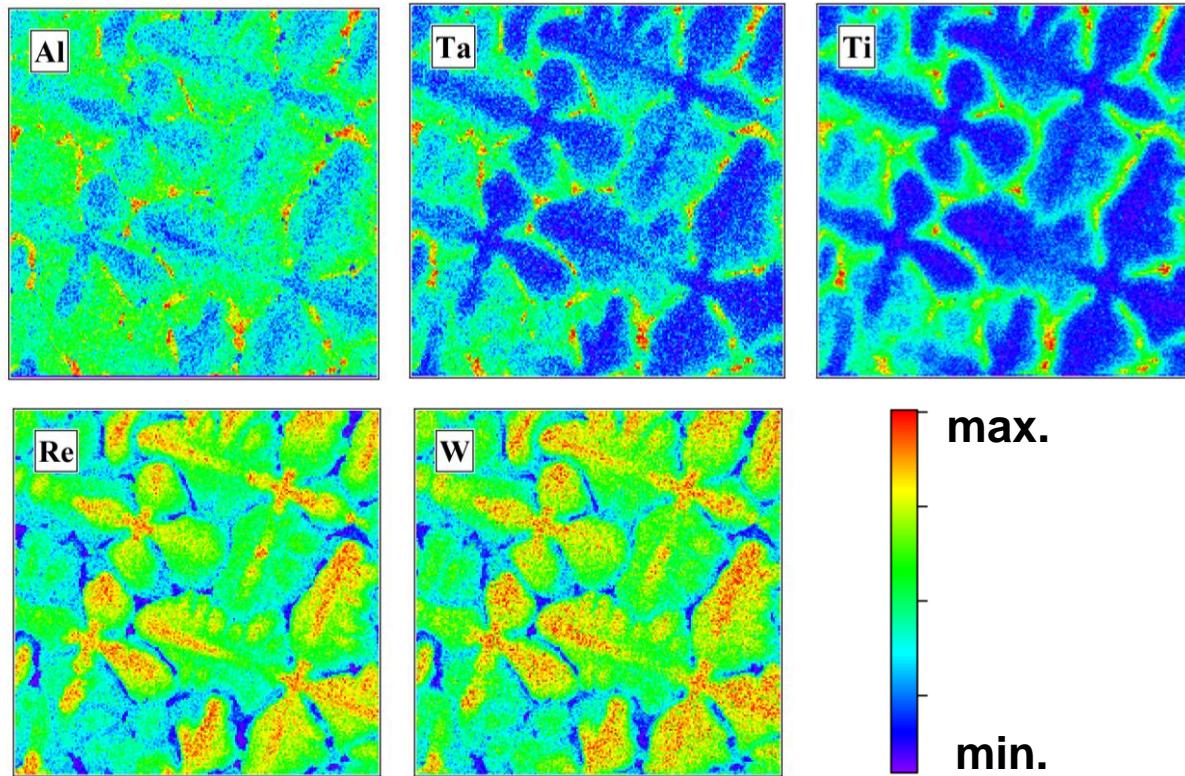
Dendrite arm spacing?



Exercise Nr.2



250 μm



Exercise Nr.2

- Spotscans / point measurements of element concentration in different regions (Values in weight pct).

Measurement#	Al	Ti	Cr	Co	Ni	Mo	Ta	W	Re
Interdendritic 1	5.69	1.07	5.99	9.63	60.56	0.72	7.54	6.01	2.77
...									
Interdendritic 5	5.85	1.18	5.77	9.51	60.38	0.72	7.8	6.4	2.4
Dendrite core 1	5.47	1.01	6.43	9.85	58.62	0.58	5.93	8.15	3.69
...									
Dendrite core 5	5.46	0.82	6.38	9.96	58.67	0.69	5.25	8.89	3.87

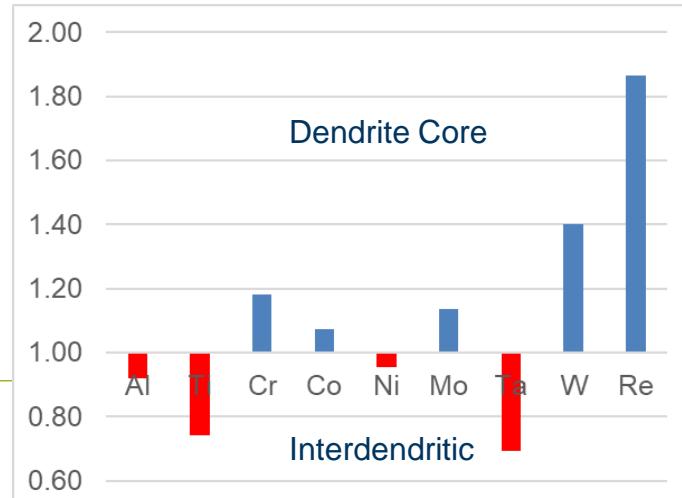
- The following relation exists:

$$k_i' = \frac{C_{D,i}}{C_{ID,i}}$$

Exercise Nr.2

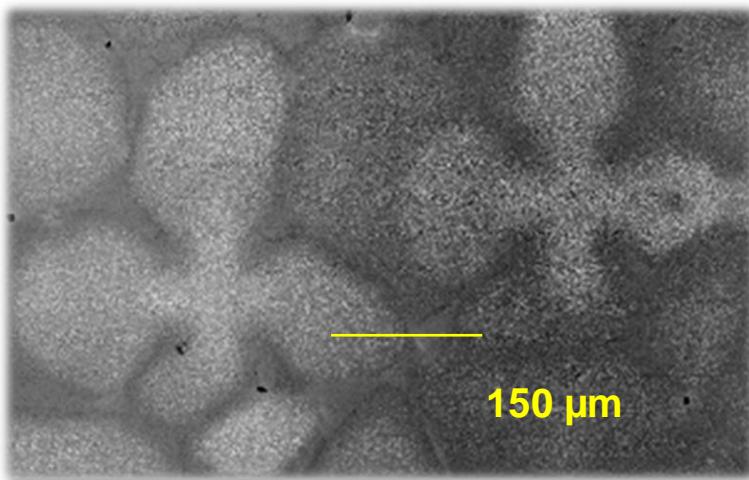
- Spotscans / point measurements of element concentration in different regions (Values in weight pct).

	Al	Ti	Cr	Co	Ni	Mo	Ta	W	Re
Interdendritic	6.01	1.20	5.29	9.14	Bal.	0.62	8.02	6.13	1.98
Dendrite core	5.53	0.89	6.24	9.81	Bal.	0.71	5.56	8.59	3.70
Segregation coefficient k_i	0.92	0.74	1.18	1.07	0.96	1.14	0.69	1.40	1.86
Nominal	5.6	1	6.4	9.7	Bal.	0.6	6.5	6.4	3



Exercise Nr.2

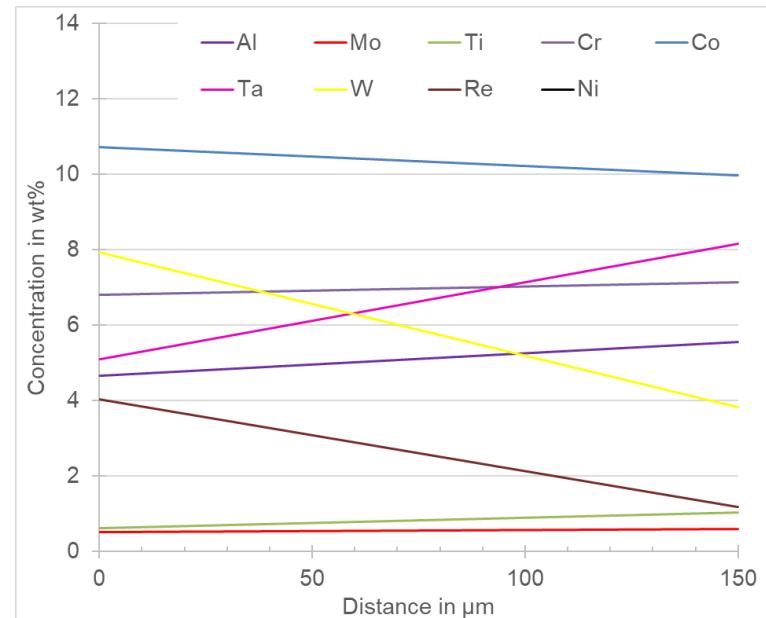
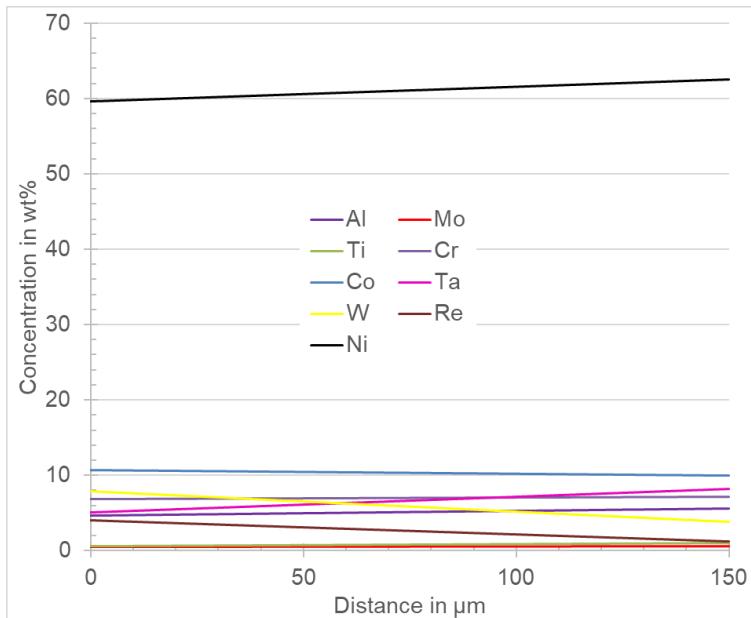
Line-scan. Raw data of detected element concentration as a function of distance of 150 µm (Values in weight pct)



Point	Distance	Al	Mo	Re	Ni
1	0	4.66	0.52		4.03	59.62
2	1	4.666	0.5205		4.011	59.64
3	2	4.672	0.521		3.992	59.659
.....
....
150	149	5.554	0.5945		1.199	62.526
151	150	5.56	0.595		1.18	62.545

Exercise Nr.2

Line-scan



Exercise Nr.3

Characteristic temperatures calculated via Thermo-Calc

Equilibrium calculations for CMSX-4 (nominal, DC and ID)

- Determination of the main transformation temperatures of the different phases that coexist in equilibrium conditions and calculation of the processing-window temperature for this alloy.
- Representation of the evolution of amount of phases with temperature.

Average values	Al	Ti	Cr	Co	Ni	Mo	Ta	W	Re
Interdendritic	6.01	1.20	5.29	9.14	Bal.	0.62	8.02	6.13	1.98
Dendrite core	5.53	0.89	6.24	9.81	Bal.	0.71	5.56	8.59	3.70
Nominal	5.6	1.0	6.4	9.7	Bal.	0.6	6.5	6.4	3.0

T in Kelvin	Liquidus	Solidus	γ' solvus
Interdendritic					
Dendrite core					
Nominal					

TASK number 3 (dendritic composition)

Thermo - Calc 2023b	State variable expression: T=1700 p=1e+05 n=1	Min value /0/:1000
SYS: SET_LOG_FILE	POLY: SET_CONDITION w(al)=0.0553	Max value /1/:1700
Heading:	..	Increment /22.5/:10
SYS:GOTO_MODULE	POLY: SET_CONDITION w(..)=...	POLY: STEP_WITH_OPTIONS
MODULE NAME: DATA	POLY: LIST_CONDITIONS	Option? /NORMAL/: NORMAL
TDB_TCFE10:SWITCH_DATABASE	POLY: COMPUTE_EQUILIBRIUM	<i>No initial equilibrium, using default</i>
DATABASE NAME /TCFE10/:TCNi10	POLY: LIST_EQUILIBRIUM	<i>Step will start from axis value 1700.00</i>
TDB_TCNI10:DEFINE_ELEMENTS	OUTPUT TO SCREEN OR FILE /SCREEN/:	...OK
ELEMENTS:Al Ti Cr Co Ni Mo Ta W Re	Options /VWCS/: (<i>see next slide</i>)
TDB_TCNI10:GET_DATA	FIRST RESULT (LIQUID)
TDB_TCNI10:GOTO_MODULE	POLY: SET_AXIS_VARIABLE	
MODULE NAME: POLY 3	Axis number: /1/:1	POLY: POST
POLY: SET_CONDITION	Condition /NONE/:T	

TASK number 3 (dendritic composition)

No initial equilibrium, using default

Step will start from axis value 1700.00

...OK

Phase Region from 1700.00 for:
LIQUID

Global check of adding phase at 1.66050E+03

Calculated 6 equilibria

Phase Region from 1677.70 for:
LIQUID
FCC_L12#1

Global check of removing phase at 1.62529E+03

Calculated 8 equilibria

Phase Region from 1625.29 for:
FCC_L12#1

Global test at 1.55000E+03 OK

Phase Region from 1522.54 for:
FCC_L12#2

Global test at 1.45000E+03 OK

Phase Region from 1064.77 for:
FCC_L12#1
FCC_L12#2
R_PHASE

SIGMA

Global test at 1.34000E+03 OK

SIGMA

Terminating at 1000.00

TASK number 3 (DC)

Thermo - Calc 2023b

SYS: POST

POST:SET_DIAGRAM_AXIS

AXIS (X, Y OR Z) :X

VARIABLE :T-C

POST:SET_DIAGRAM_AXIS

AXIS (X, Y OR Z) :y

VARIABLE :VPV(*)

COLUMN NUMBER /*/*

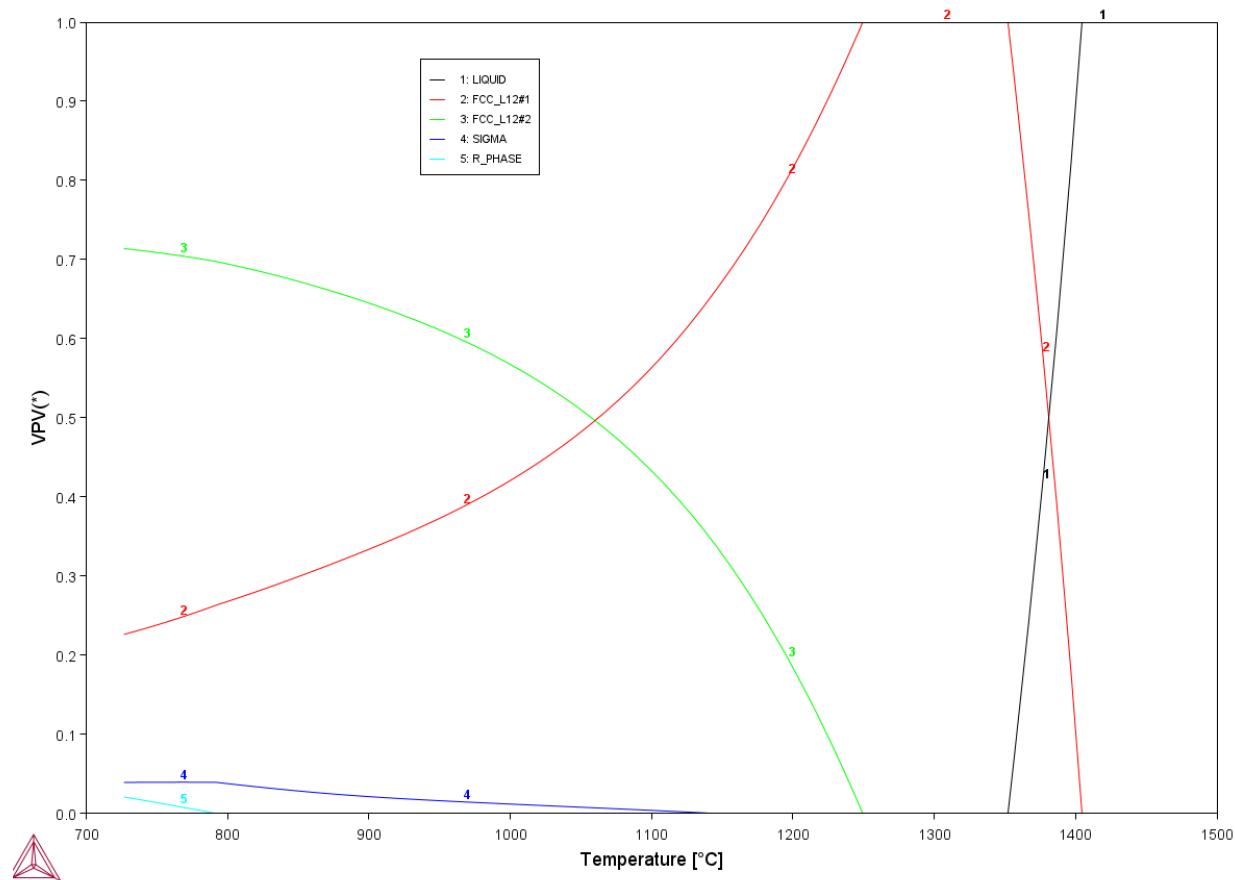
POST:SET_LABEL_CURVE_OPTION

CURVE LABEL OPTION (A, B, C, D, E, F OR N) /N:/F

POST:PLOT_DIAGRAM

POST:MAKE_EXPERIMENTAL_DATAFILE

OUTPUT TO SCREEN OR FILE /SCREEN/:FILE



TASK number 3 (Interdendritic composition)

Thermo - Calc 2023b	State variable expression: T=1700 p=1e+05 n=1	Min value /0/:1000
SYS: SET_LOG_FILE	POLY: SET_CONDITION w(al)=0.0601	Max value /1/:1700
Heading:	..	Increment /22.5/:10
SYS:GOTO_MODULE	POLY: SET_CONDITION w(..)=...	POLY: STEP_WITH_OPTIONS
MODULE NAME: DATA	POLY: LIST_CONDITIONS	Option? /NORMAL/: NORMAL
TDB_TCFE10:SWITCH_DATABASE	POLY: COMPUTE_EQUILIBRIUM	<i>No initial equilibrium, using default</i>
DATABASE NAME /TCFE10/:TCNi10	POLY: LIST_EQUILIBRIUM	<i>Step will start from axis value 1700.00</i>
TDB_TCNI10:DEFINE_ELEMENTS	OUTPUT TO SCREEN OR FILE /SCREEN/:	...OK
ELEMENTS:Al Ti Cr Co Ni Mo Ta W Re	Options /VWCS/: (<i>see next slide</i>)
TDB_TCNI10:GET_DATA	FIRST RESULT (LIQUID)
TDB_TCNI10:GOTO_MODULE	POLY: SET_AXIS_VARIABLE	
MODULE NAME: POLY 3	Axis number: /1/:1	POLY: POST
POLY: SET_CONDITION	Condition /NONE/:T	

TASK number 3 (Interdendritic composition)

No initial equilibrium, using default

Phase Region from 1608.05 for:

Step will start from axis value 1700.00

FCC_L12#1

Phase Region from 1189.22 for:

FCC_L12#1

...OK

Global test at 1.53000E+03 .. Back....

FCC_L12#2

Phase Region from 1700.00 for:

SIGMA

LIQUID

Phase Region from 1577.16 for:

Global test at 1.11000E+03 O

Global check of adding phase at 1.66050E+03

FCC_L12#1

Calculated 6 equilibria

FCC_L12#2

Phase Region from 1106.45 for:

Phase Region from 1660.50 for:

Global test at 1.50000E+03 OK

FCC_L12#1

LIQUID

FCC_L12#2

FCC_L12#1

R_PHASE

Global check of removing phase at 1.60805E+03

SIGMA

Calculated 9 equilibria

Global test at 1.03000E+03 OK

Terminating at 1000.00

TASK number 3 (ID)

Thermo - Calc 2023b

SYS: POST

POST:SET_DIAGRAM_AXIS

AXIS (X, Y OR Z) :X

VARIABLE :T-C

POST:SET_DIAGRAM_AXIS

AXIS (X, Y OR Z) :y

VARIABLE :VPV(*)

COLUMN NUMBER /*/*

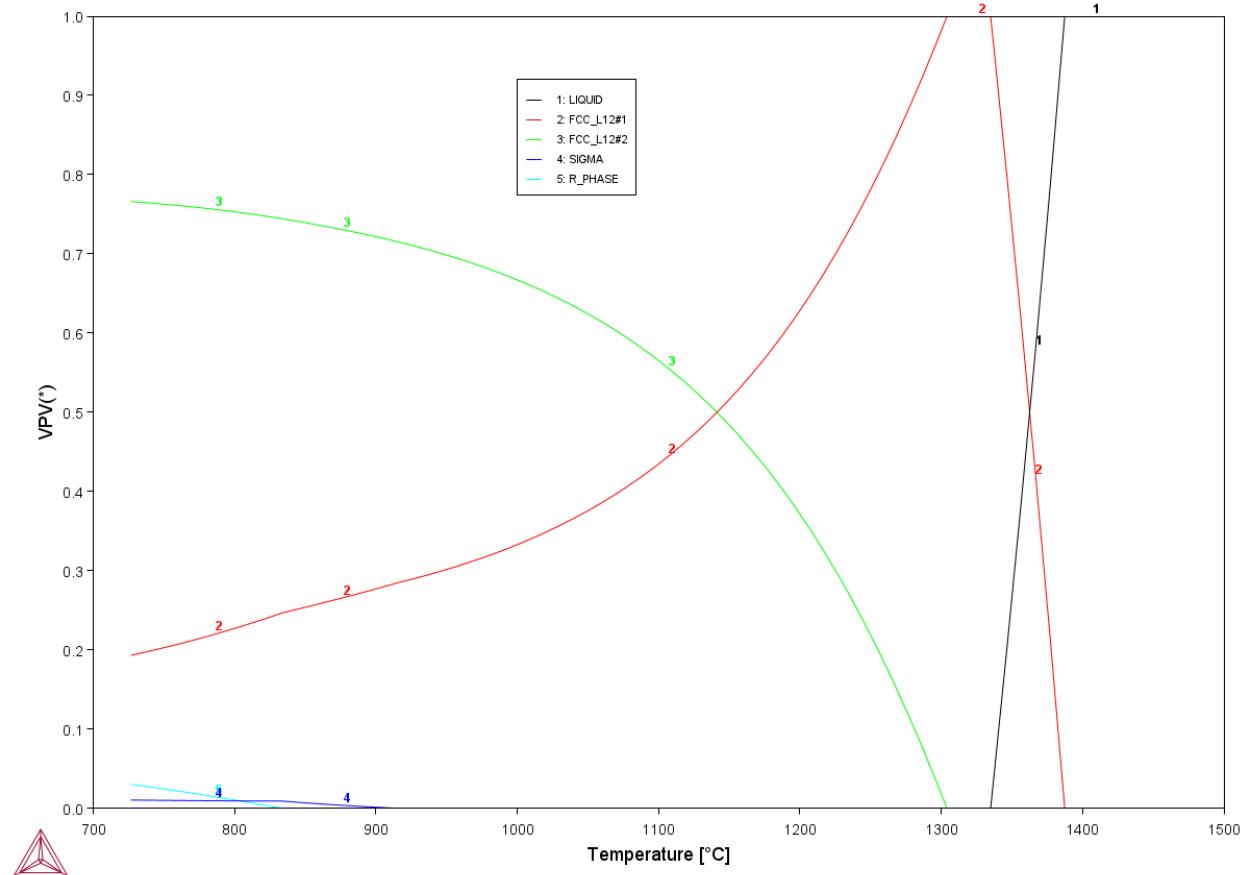
POST:SET_LABEL_CURVE_OPTION

CURVE LABEL OPTION (A, B, C, D, E, F OR N) /N:/F

POST:PLOT_DIAGRAM

POST:MAKE_EXPERIMENTAL_DATAFILE

OUTPUT TO SCREEN OR FILE /SCREEN/:FILE



TASK number 3 (Nominal composition)

Thermo - Calc 2023b	State variable expression: T=1700 p=1e+05 n=1	Min value /0/:1000
SYS: SET_LOG_FILE	POLY: SET_CONDITION w(al)=0.0560	Max value /1/:1700
Heading:	..	Increment /22.5/:10
SYS:GOTO_MODULE	POLY: SET_CONDITION w(..)=...	POLY: STEP_WITH_OPTIONS
MODULE NAME: DATA	POLY: LIST_CONDITIONS	Option? /NORMAL/: NORMAL
TDB_TCFE10:SWITCH_DATABASE	POLY: COMPUTE_EQUILIBRIUM	<i>No initial equilibrium, using default</i>
DATABASE NAME /TCFE10/:TCNi10	POLY: LIST_EQUILIBRIUM	<i>Step will start from axis value 1700.00</i>
TDB_TCNI10:DEFINE_ELEMENTS	OUTPUT TO SCREEN OR FILE /SCREEN/:	...OK
ELEMENTS:Al Ti Cr Co Ni Mo Ta W Re	Options /VWCS/: (<i>see next slide</i>)
TDB_TCNI10:GET_DATA	FIRST RESULT (LIQUID)
TDB_TCNI10:GOTO_MODULE	POLY: SET_AXIS_VARIABLE	
MODULE NAME: POLY 3	Axis number: /1/:1	POLY: POST
POLY: SET_CONDITION	Condition /NONE/:T	

TASK number 3 (Nominal composition)

No initial equilibrium, using default	Phase Region from 1620.86 for:	Phase Region from 1206.81 for:
Step will start from axis value 1700.00	FCC_L12#2	FCC_L12#1
...OK	Global test at 1.55000E+03 OK	FCC_L12#2
Phase Region from 1700.00 for:		SIGMA
LIQUID		Global test at 1.13000E+03 OK
Global check of adding phase at 1.67173E+03	Phase Region from 1532.85 for:	
Calculated 5 equilibria	FCC_L12#1	Phase Region from 1020.45 for:
Phase Region from 1671.73 for:	FCC_L12#2	FCC_L12#1
LIQUID	Global test at 1.46000E+03 OK	FCC_L12#2
FCC_L12#2		R_PHASE
Global check of removing phase at 1.62086E+03		SIGMA
Calculated 8 equilibria		Terminating at 1000.00

TASK number 3 (Nominal composition)

Thermo - Calc 2023b

SYS: POST

POST:SET_DIAGRAM_AXIS

AXIS (X, Y OR Z) :X

VARIABLE :T-C

POST:SET_DIAGRAM_AXIS

AXIS (X, Y OR Z) :y

VARIABLE :VPV(*)

COLUMN NUMBER /*:/*

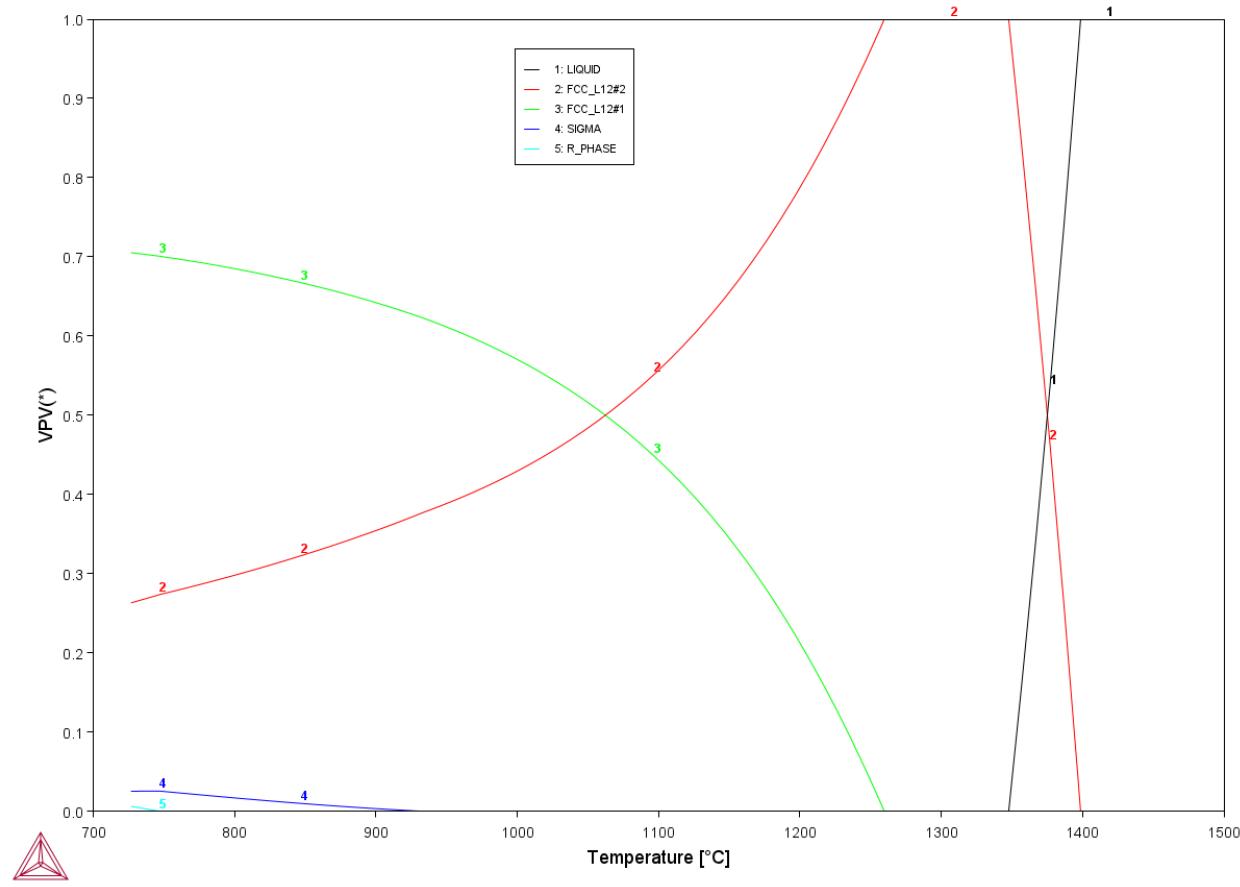
POST:SET_LABEL_CURVE_OPTION

CURVE LABEL OPTION (A, B, C, D, E, F OR N) /N:/F

POST:PLOT_DIAGRAM

POST:MAKE_EXPERIMENTAL_DATAFILE

OUTPUT TO SCREEN OR FILE /SCREEN/:FILE



Exercise Nr.3

Characteristic temperatures calculated via Thermo-Calc

Equilibrium calculations for CMSX-4 (nominal, DC and ID)

- a. Determination of the main transformation temperatures of the different phases that coexist in equilibrium conditions and calculation of the processing-window temperature for this alloy.
- b. Representation of the evolution of amount of phases with temperature.

T in Kelvin	Liquidus	Solidus	γ' solvus	Sigma	R_Phase
Interdendritic	1660.50	1608.05	-	1189.22	1106.45
Dendrite core	1677.70	1625.29	1522.54	1417.93	1064.77
Nominal	1671.73	1620.86	1532.85	1206.81	1020.45

Exercise Nr.3

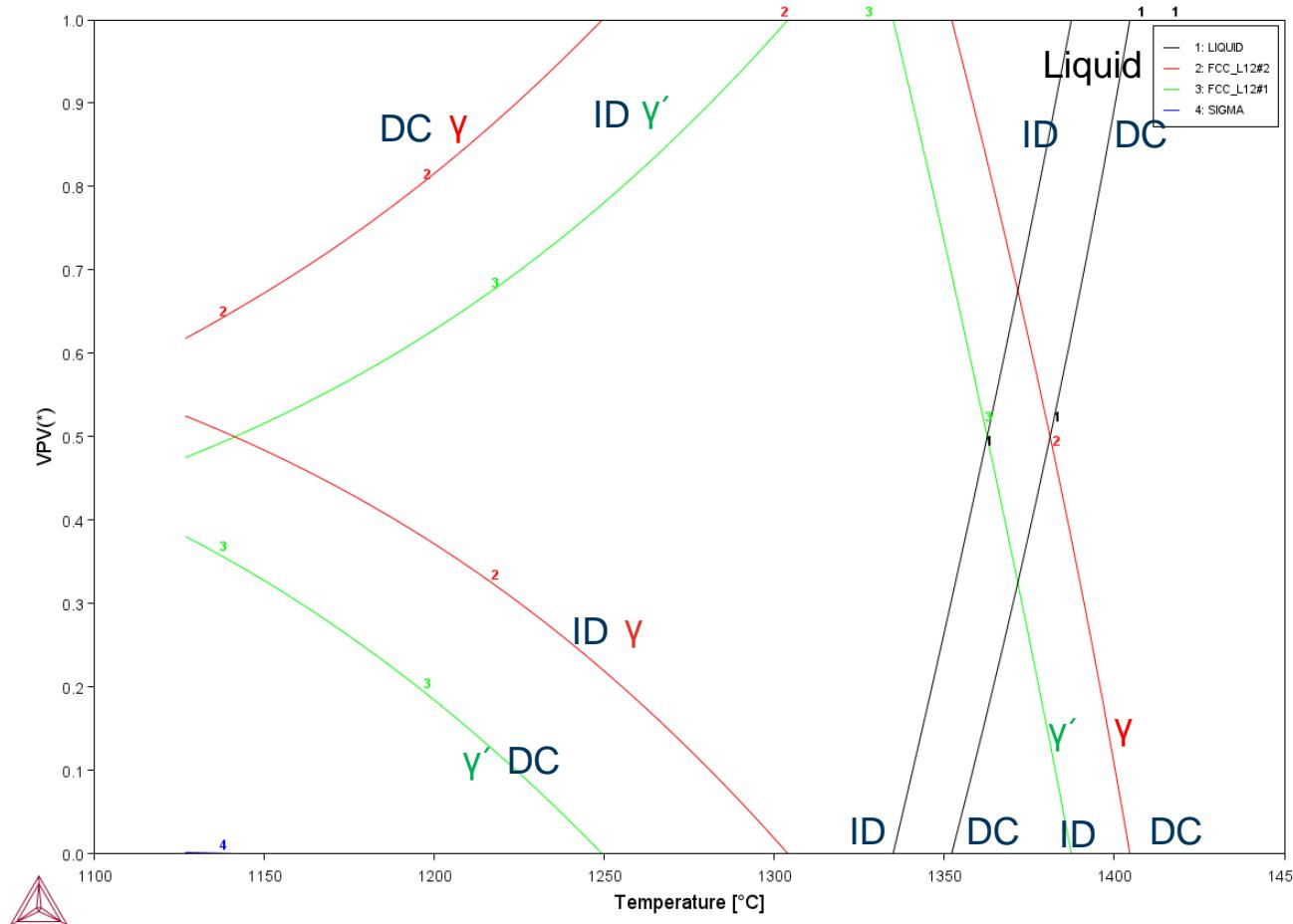
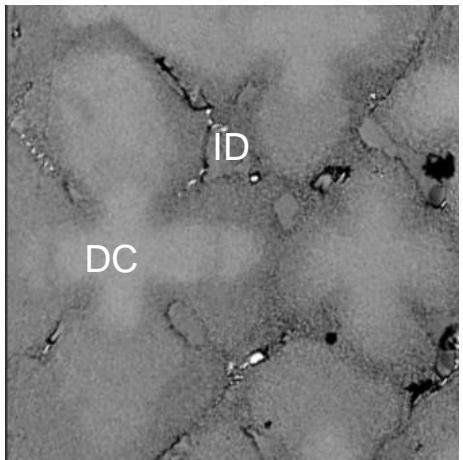
Characteristic temperatures calculated via Thermo-Calc

Equilibrium calculations for CMSX-4 (nominal, DC and ID)

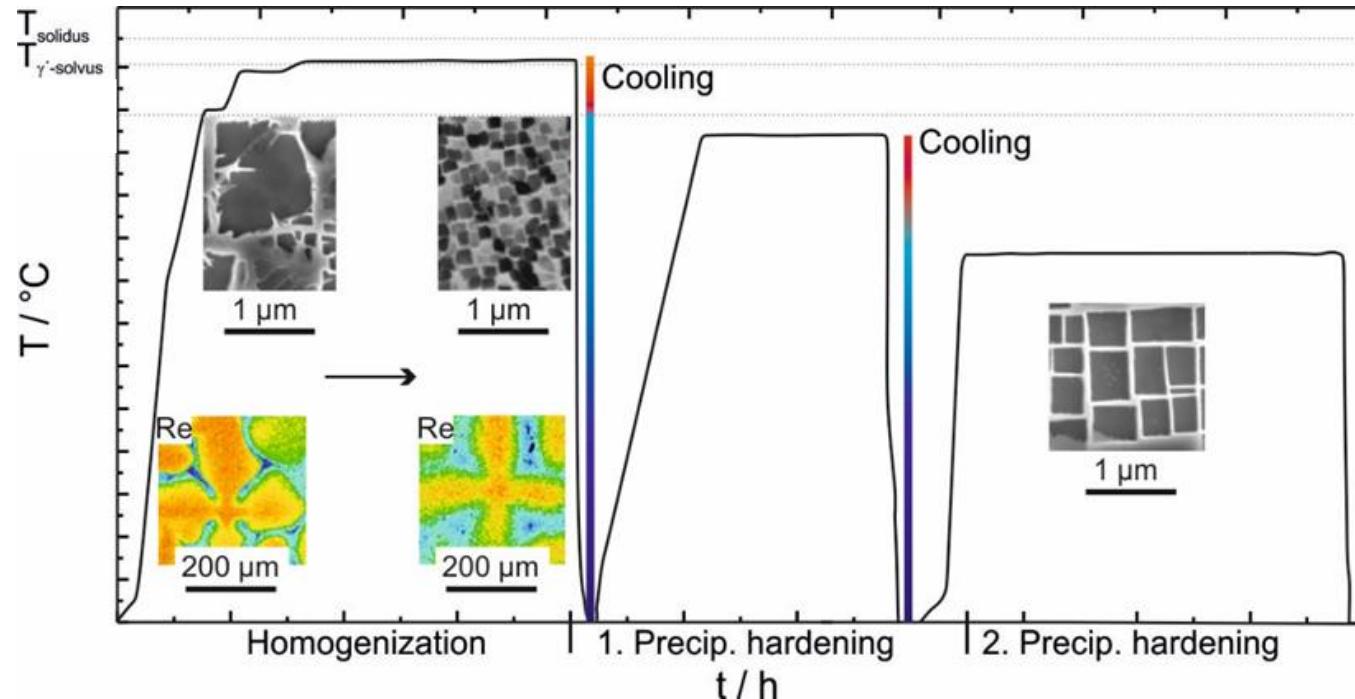
- a. Determination of the main transformation temperatures of the different phases that coexist in equilibrium conditions and calculation of the processing-window temperature for this alloy.
- b. Representation of the evolution of amount of phases with temperature.

T in °C	Liquidus	Solidus	γ' solvus	Sigma	R_Phase
Interdendritic	1387.35	1334.9	-	916.07	833.3
Dendrite core	1404.55	1352.14	1249.39	1144.78	791.62
Nominal	1398.58	1347.71	1259.7	933.66	747.3

Exercise Nr.3

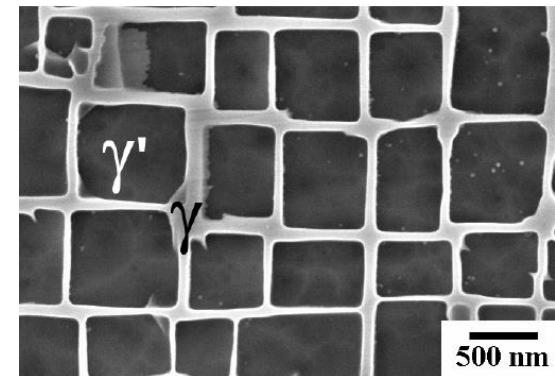


Conventional processing route



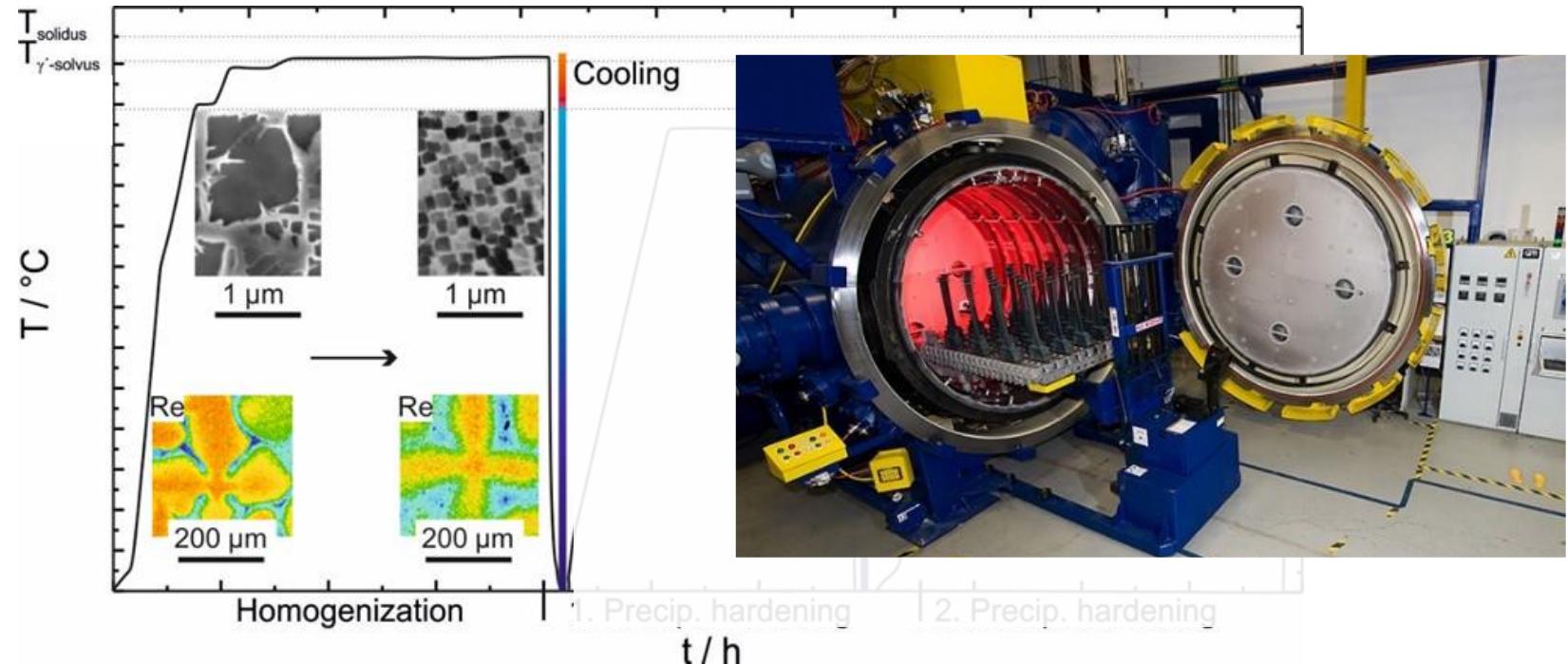
Conventional processing route

- Objective:
 - (Macroscopically) homogeneous element distribution
 - Uniform fine γ' - precipitates within the γ -matrix (γ -channels)
- How?:
 - **Solution and homogenization of the γ -field**
 - => Dissolution of the γ'
 - => Homogenization by diffusion
 - Fast cooling / Quenching
 - **Precipitation hardening / Aging**

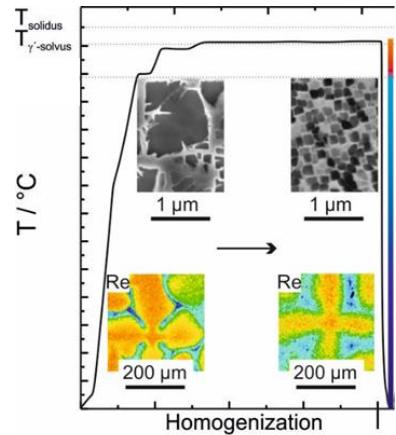


- Very long heat-treatments (€€€)
- Increase of porosity (Kirkendall)

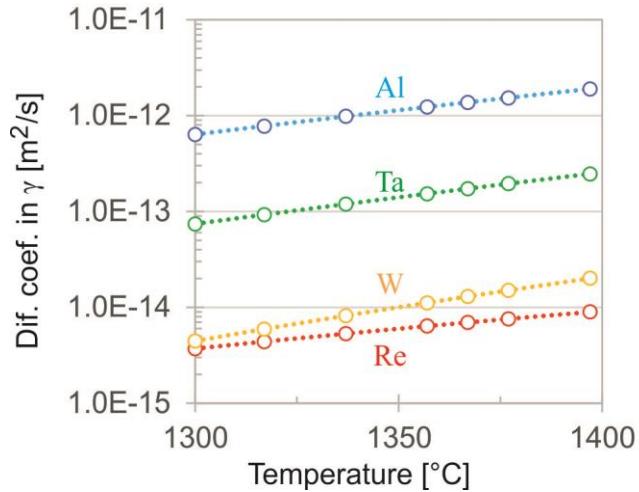
Conventional processing route



Conventional processing route

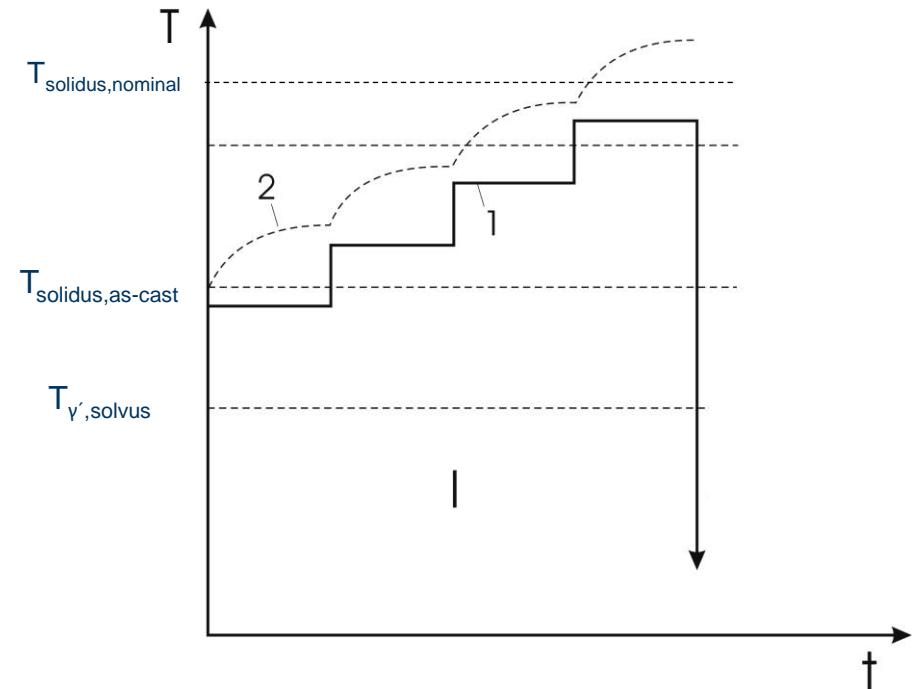


- Very long heat-treatments (€€€)
- Increase of porosity (Kirkendall)
- Deterioration of mech. properties

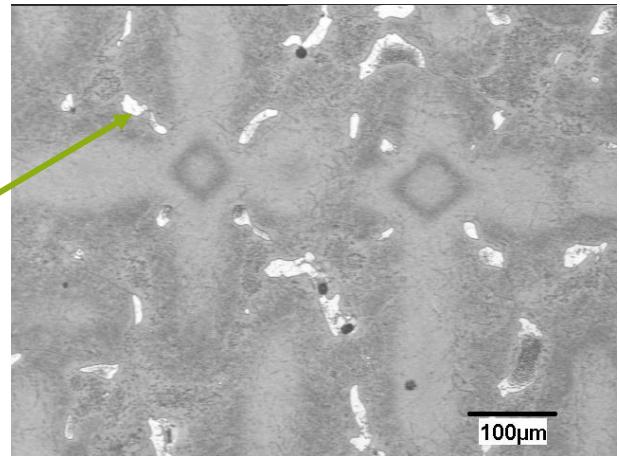


Complex SXs are no longer attractive for the aeronautical sector despite their potential

Solution annealing “Stepwise”

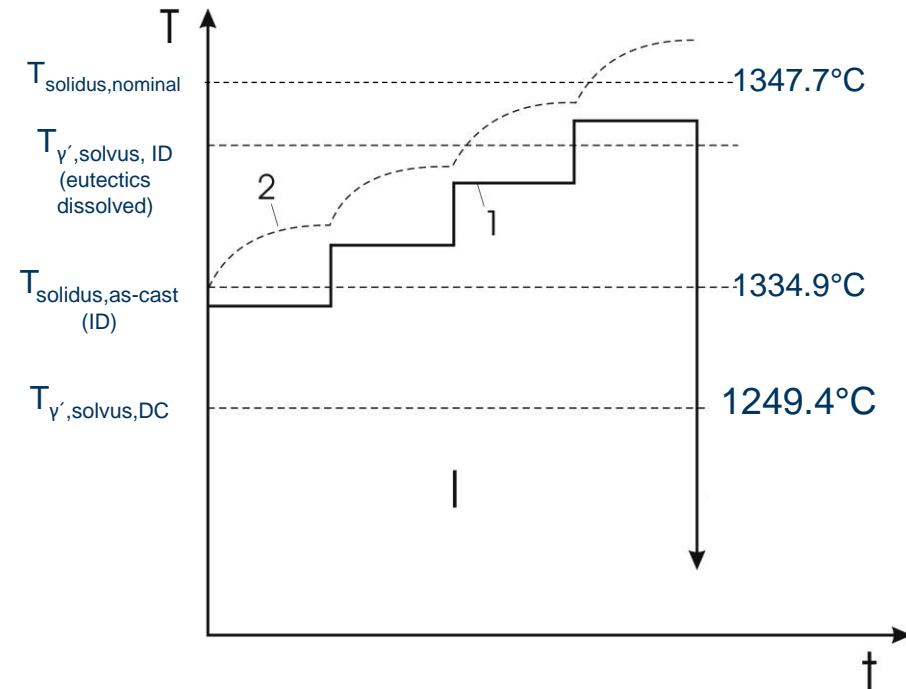


„incipient melting“



- Strong segregation lead to locally different T_{solidus}
- Sometimes $T_{\text{solidus,lokal}} < T_{\gamma'-\text{solvus,global}}$

Solution annealing “Stepwise”



T in °C	Solidus	γ' solvus
Interdendritic	1334.9	-
Dendrite core	1352.14	1249.39
Nominal	1347.71	1259.7

- Strong segregation lead to locally different T_{solidus}
- Sometimes $T_{\text{solidus,lokal}} < T_{\gamma'-\text{solvus, ID}}$

Exercise Nr.4

Interdiffusion coefficients in γ - phase

- a. Calculation of interdiffusion coefficients of the alloying elements in the γ - phase at 1300°C.
- b. Which alloying element determines the duration of the solution annealing heat treatment? Which alloying element will be the first to homogenize?

Exercise Nr.4

Thermo - Calc 2023b

SYS: SET_LOG_FILE	TDB_TCNI10:GET_DATA	State variable expression: T=1573.15 p=1e+05 n=1
Heading:	TDB_TCNI10:APPEND	POLY: SET_CONDITION w(al)=0.0560
SYS:GOTO_MODULE	DATABASE NAME /TCNI10/:MOBNi5	..
MODULE NAME: DATA	APP:DEFINE_SYSTEM	POLY: SET_CONDITION w(..)=...
TDB_TCFE10:SWITCH_DATABASE	ELEMENTS:Ni Co Cr Al Mo Ti Re W Ta	POLY: LIST_CONDITIONS
DATABASE NAME /TCFE10/:TCNi10	APP:GET_DATA	POLY: COMPUTE_EQUILIBRIUM
TDB_TCNI10:DEFINE_SPECIES	TDB_TCNI10:GOTO_MODULE	POLY: LIST_EQUILIBRIUM
SPECIES:Ni Co Cr Al Mo Ti Re W Ta	MODULE NAME: POLY 3	OUTPUT TO SCREEN OR FILE /SCREEN/:
	POLY: SET_CONDITION	Options /VWCS/:
		POLY: show-val DT(FCC_L12,Ti)
		DT(FCC_L12,Ti)=1.1864672E-13

Exercise Nr.4

DT(FCC_L12, TI)=1.1864672E-13

[m²/s]

DT(FCC_L12, AL)=3.1910091E-13

DT(FCC_L12, TA)=6.0495126E-14

DT(FCC_L12, W)=1.0794955E-14

DT(FCC_L12, RE)=2.1409152E-15

DT(FCC_L12, MO)=5.7555214E-14

DT(FCC_L12, CR)=1.9909263E-13

DT(FCC_L12, CO)=8.6006041E-14

DT(FCC_L12, NI)=2.8196456E-14

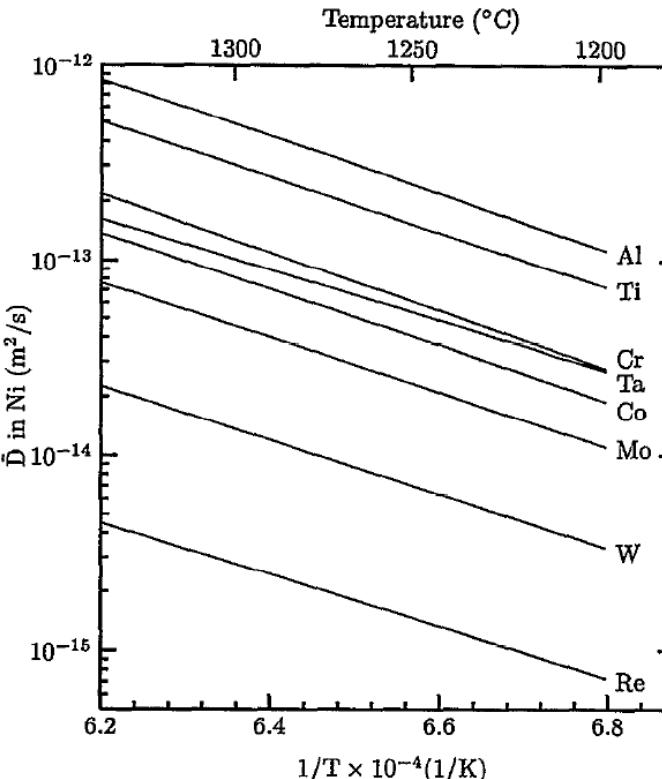


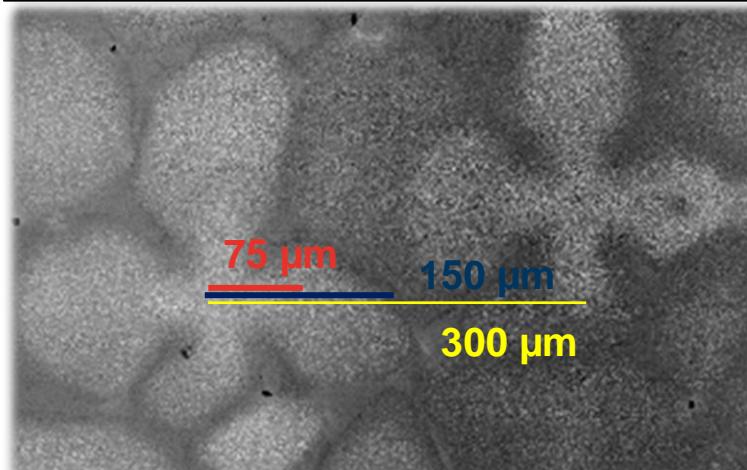
Figure 1: Variation of the interdiffusion coefficient in pure nickel of various elements pertinent to the superalloys, as a function of inverse temperature.

Exercise Nr.4

Assumption: Homogenization is reached when the diffusion length, d_{diff} , equals one half of the dendrite arm spacing i.e., the distance between maximum and minimum of the element concentration.

Hint: You've already measured the distance. Take a closer look onto the units of D.

D is the diffusion coefficient of a solute in free solution. The diffusion coefficient determines the time it takes a solute to diffuse a given distance in a medium. D has the units of area/time (typically cm^2/s). Its value is unique for each solute and must be determined empirically.



$$DT(\text{FCC_L12, RE}) = 2.14 \text{ E-15}$$

$$\text{Units of D: } [\text{m}^2/\text{s}] = (\text{Distance})^2/t$$

$$t = (\text{Distance})^2/D$$

$$\text{Distance} = PDS/4 = 75 \mu\text{m} = 7.5 \text{ E-5 m} = 0.000075 \text{ m}$$

$$t = (0.000075)^2 / 2.14 \text{ E-15} = 2.63E+6 \text{ s} = 730,6 \text{ h} > 30,44 \text{ days}$$

Exercise Nr.5

Simulation of the solution annealing heat treatment by using Dictra. For this task, the composition of the simplified alloy listed in Table 1 will be used.

- a. Isothermal homogenization heat treatments at 1400°C. Homogenization time = 1h.
- b. For each isothermal heat treatment, graphical representation of the homogenization profiles for the different homogenization steps.
- c. Calculation of the time necessary to achieve homogenization at different temperatures.

Table 1. Nominal chemical compositions in wt.% of the two alloys used in the exercise: a complex single crystal Ni base superalloy, CMSX-4, and a simplified alloy, S4_alloy.

	Cr	Co	Mo	W	Ta	Ti	Al	Hf	Re	Ni
CMSX-4	6.5	9	0.6	6	6.5	1	5.6	0.1	3	Bal
S4_alloy	6.5	9	-	-	-	-	5.6	-	-	Bal

Exercise Nr.5

Thermo - Calc 2023b

SYS: SET_LOG_FILE

Heading:

SYS:GOTO_MODULE

MODULE NAME: DATA

TDB_TCFE10:SWITCH_DATABASE

DATABASE NAME /TCFE10/:TCNi10

TDB_TCNI10: DEFINE_SYSTEM

ELEMENTS:Ni Co Cr Al

TDB_TCNI10: REJECT PHASES *

TDB_TCNI10: RESTORE PHASES FCC_L12

TDB_TCNI10: GET

TDB_TCNI10:APPEND

DATABASE NAME /TCNI10/:MOBNi5

... DEFINE_SYSTEM

.... Ni Co Cr Al

.... REJECT PHASES *

... RESTORE PHASES FCC_L12

... GET

TDB_TCNI10: :GOTO_MODULE

MODULE NAME: DICTRA_MONITOR

DIC> INPUT_SCHEIL_PROFILE → TXT file

ENTER WIDTH OF REGION /1/:150e-6

PHASE NAME:fcc_l12#1

SHOULD MORE PHASES BE ENTERED IN THE REGION /NO/:n

DIC> SET_CONDITION

GLOBAL OR BOUNDARY CONDITION /GLOBAL/:glob

VARIABLE :T

LOW TIME LIMIT /0/:0

T(TIME,X)=1673.15;

HIGH TIME LIMIT /*:*

ANY MORE RANGES /N:/N

DIC>SET_SIMULATION_TIME

END TIME FOR INTEGRATION /.1/:3600

AUTOMATIC TIMESTEP CONTROL /YES/:y

MAX TIMESTEP DURING INTEGRATION /360/:10

INITIAL TIMESTEP :/1E-07/:

SMALLEST ACCEPTABLE TIMESTEP :/1E-07/:

DIC>SAVE_WORKSPACES

DIC>SET_INTERACTIVE

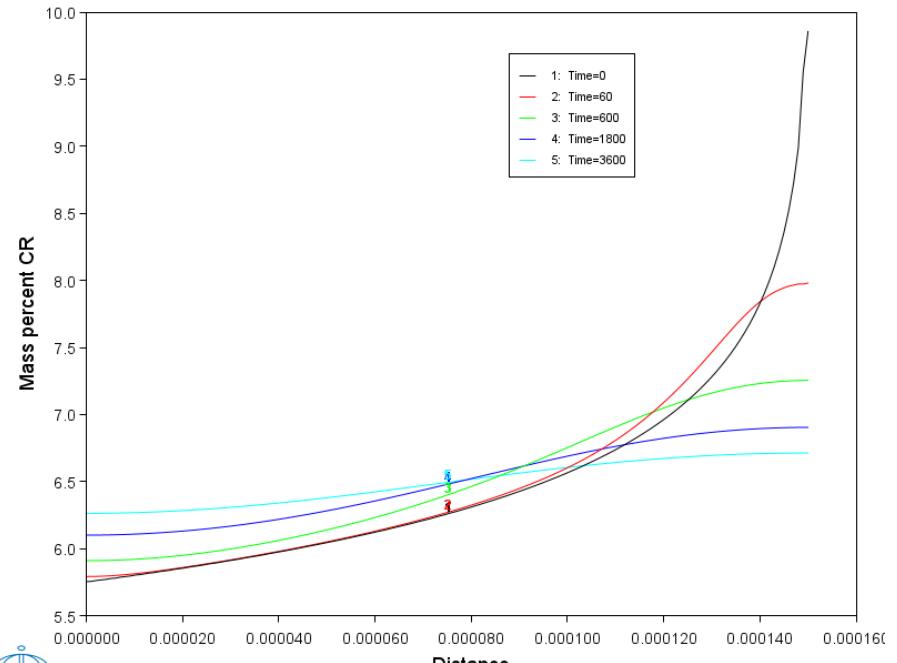
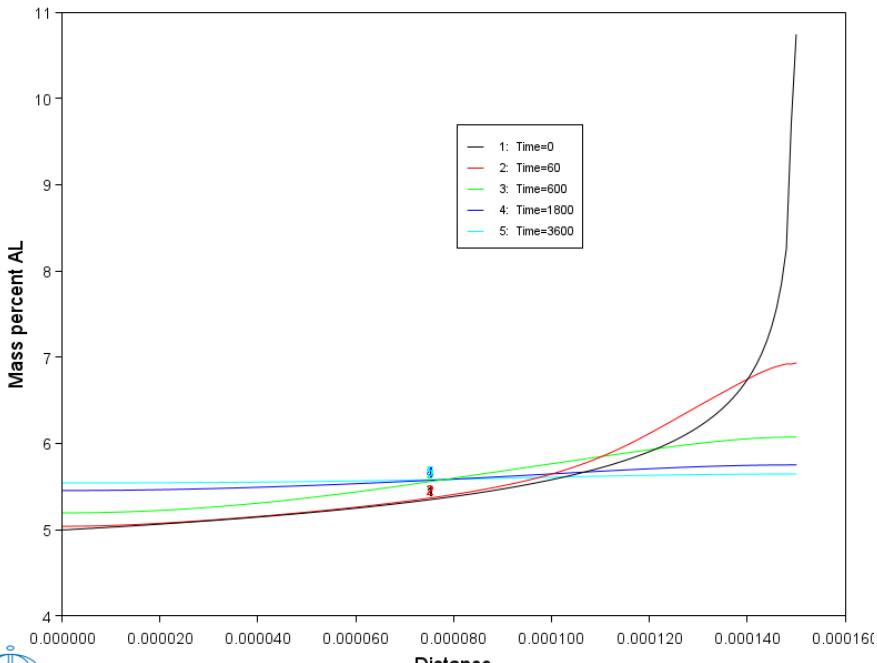
DIC>SIMULATE_REACTION

Exercise Nr.5

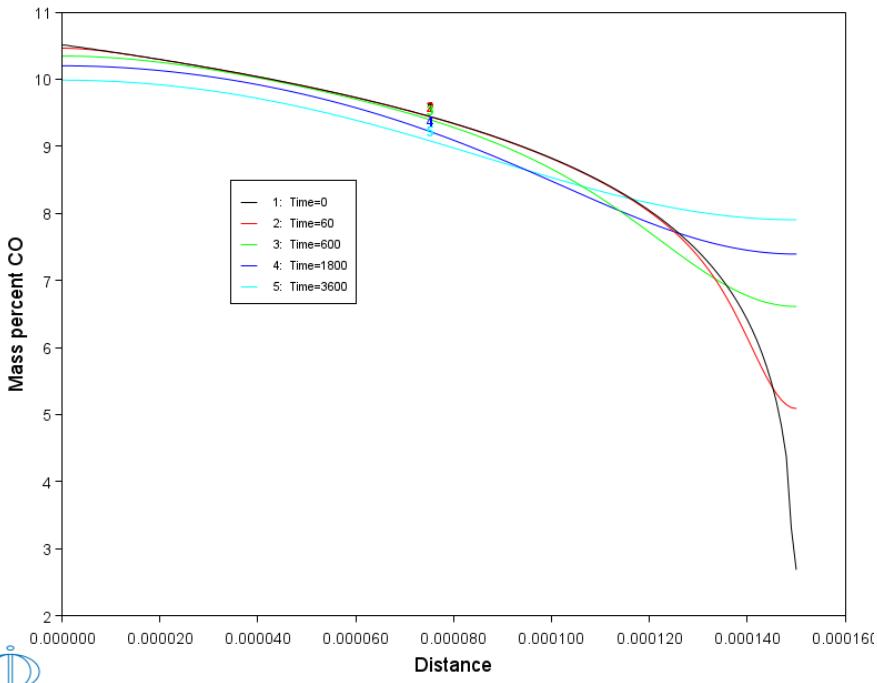
Thermo - Calc 2023b

	s-d-a	s-d-a
Post	y	y
s-p-c	w-p(FCC_L12#1)	w-p(FCC_L12#1)
time	al	cr
0,60,120,180,240,300,360,420,48 0,540,600,1800,3600	plot make	plot make
label-curv	file	file
y	s-d-a	s-d-a
s-d-a	y	y
x	w-p(FCC_L12#1)	w-p(FCC_L12#1)
distance	co	ni
global	plot make	Plot make
	file	file

Exercise Nr.5



Exercise Nr.5

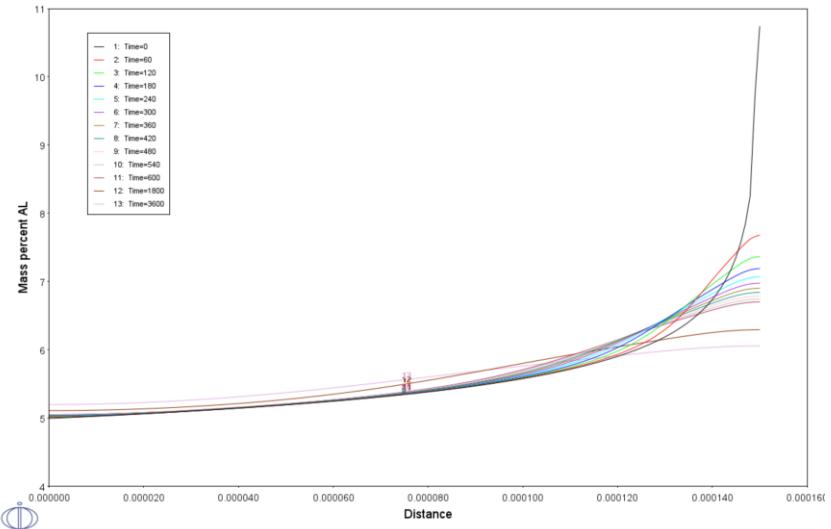


Exercise Nr.5

Diagrams: Al

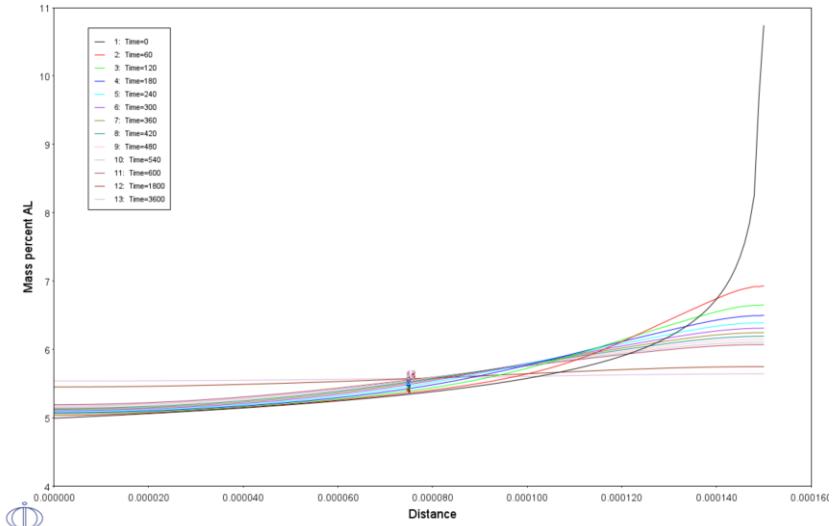
@ 1250 °C up to 1h

2021.10.04.13.10.52
Time = 0,60,120,180,240,300,360,420,480,540,600,1800,3600
CELL #1



@ 1400 °C up to 1h

2021.10.04.11.47.20
Time = 0,60,120,180,240,300,360,420,480,540,600,1800,3600
CELL #1



Überprüfungsfragen

1. Welchen Belastungen müssen Superlegierungen standhalten können und durch welche Legierungstechnischen Maßnahmen wird dies sichergestellt?
2. Wie wird im Feinguss die Erstarrung als Einkristall erreicht? Wodurch wird die Kristallorientierung beeinflusst?
3. Welche Gussparameter beeinflussen das Erstarrungsgefüge?
4. Nennen Sie die typischen Legierungselemente einkristalliner Ni-basis Superlegierungen der 2. Generation. Was ist der Grund für die inhomogene Elementverteilung?
5. Warum liegt eine kubische bzw. würzelförmige Morphologie der γ' -Phase vor?
6. Wie ist die Porosität mit Blick auf die mechanischen Eigenschaften zu beurteilen?
7. Erklären Sie mit einem Schema den Verlauf (Temperatur vs. Zeit) der Wärmebehandlung von einer Nickelbasis Superlegierung nach der Erstarrung.
8. Warum ist das Lösungsglühen von Superlegierungen nötig? Was ist das größte Risiko dieser Wärmebehandlung?
9. Welchen Einfluss hat die Abkühlgeschwindigkeit vom Lösungsglühen?

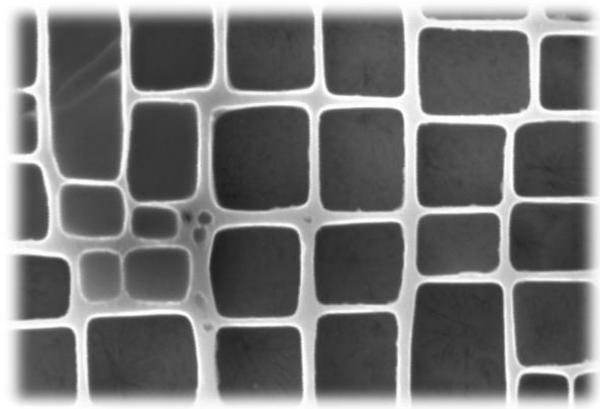
Literature

- Reed, RC (2008) The Superalloys: Fundamentals and Applications. Cambridge University Press, Cambridge.
- R. Burgel (2006) Handbuch Hochtemperatur-Werkstofftechnik: Grundlagen, Werkstoffbeanspruchungen, Hochtemperaturlegierungen. Vieweg+Teubner Verlag.
- Lechner C, Seume J (2010) Stationare Gasturbinen. Springer, Berlin & Heidelberg.
- Parsa AB et. al (2015) Advanced Scale Bridging Microstructure Analysis of Single Crystal Ni-Base Superalloys. *Adv. Eng. Mater.* 17(2):216–230.
- Ruttert B, Meid C, Mujica Roncery L, Lopez-Galilea I, Bartsch M, Theisen W (2018) Effect of porosity and eutectics on the high-temperature low-cycle fatigue performance of a nickel-base single-crystal superalloy. *Scripta Mater.* 155:139–143.

Thank you for your attention !

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